

Calibration error analysis of inertially stabilized platforms using quaternions and octonions in rotation decomposition

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Abstract

In the calibration process of the inertially stabilized platforms with a high-precision turntable and an autocollimator, significant calibration errors can result from the axis misalignments between the inertially stabilized platforms and the turntable. Based on the relationship between spatial rotations and quaternions or octonions, this article proposes a representation using octonions to realize the decomposition of the rotation axis in two perpendicular axes and subsequently derives the calibration error model. The test results demonstrated that the error is significantly improved after compensation. The azimuth variance is reduced from $0.1379(^{\circ})^2$ to $0.0492(^{\circ})^2$, which offers a more accurate set of data for further compensation based on the error model of the platform itself.

Keywords

Rotation, quaternion, octonion, spinor, inertially stabilized platform calibration

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Introduction

As a vector rotates around a straight line in a threedimensional (3D) space coordinate system, it is easy to obtain the rotated vector through a rotation matrix.^{1–5} Inertially stabilized platforms (ISPs) have been widely utilized to maintain its sensor's orientation pointing to an accurate direction in many applications, such as vehicles, ships, aircrafts and spacecraft. Its accuracy is technically vital and affects the capture, location and tracking of the target.⁶⁻⁹ A two-axial platform is fixed on a high-precision turntable, with an autocollimator showing the rotating angular error,^{10,11} as shown in Figure 1, where A_1 is the azimuth axis of the turntable, B_1 its elevation axis, A_2 the azimuth axis of the platform, B₂ its elevation axis and C₂ the line of sight. At zero positions, A_1 and A_2 are vertical, while B_1 and B_2 are horizontal.

In the calibration process, θ_{PA} and θ_{PE} are the platform rotation angles in terms of azimuth and elevation, respectively. The azimuth and elevation angles can be similarly defined for the turntable (θ_{TA} and θ_{TE}) and the autocollimator readouts of axes (δ_A and δ_E). Generally, it is $\theta_{TA} = -\theta_{PA}$ and $\theta_{TE} = -\theta_{PE}$. If $\theta_{PE} = 0$ and $\theta_{TE} = 0$, it is obvious that $\delta_A = \theta_{PA} - \theta_{TA}$. However, if $\theta_{TE} = -\theta_{PE} \neq 0$, A₁ and A₂ are no longer vertical, which indicates $\delta_A \neq \theta_{PA} - \theta_{TA}$. It means some calibration error is introduced, and it is obligatory to determine the error to calibrate ISPs accurately.

This article first presents the notation and definitions of quaternions and octonions and then gives a detailed description of the relationship between rotations and quaternions or octonions. It obtains the spinor with octonions as a vector rotates around an axis by means of decomposing the shaft into two perpendicular axes, which indicates octonions can also be used in rotations. The actual test results have demonstrated that the calibration process error can be effectively compensated.

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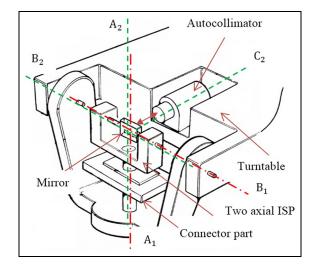


Figure I. Calibration system of ISPs.

Quaternions and octonions

Notation and definitions of quaternions

Quaternions are invented by W.R. Hamilton, and they can be used to speed up calculations involving rotations. A quaternion is represented by just four scalars, in contrast to a 3×3 rotation matrix which has nine scalar entries.¹² A quaternion can be defined as¹³

$$\mathbf{Q} = \mathbf{a}\mathbf{i} + \mathbf{b}\mathbf{j} + \mathbf{c}\mathbf{k} + \mathbf{q}_0 \tag{1}$$

where a, b, c and q_0 are real numbers, and *i*, *j* and *k* are the imaginary units which obey the following multiplication rules

$$\begin{cases} i^{2} = j^{2} = k^{2} = -1 \\ ij = -ji = k \\ ki = -ik = j \\ jk = -kj = i \end{cases}$$

Let q = ai + bj + ck, equation (1) can then be rewritten as

$$\mathbf{Q} = \mathbf{q}_0 + \boldsymbol{q} \tag{2}$$

where q_0 denotes the scalar part, and q denotes the vector part.

Given another quaternion P (P = $p_0 + p$), their product according to the algebraic rules of multiplications given above is¹³

$$PQ = (p_0 + \boldsymbol{p})(\boldsymbol{q}_0 + \boldsymbol{q})$$

= $p_0q_0 - \boldsymbol{p} \cdot \boldsymbol{q} + p_0\boldsymbol{q} + q_0\boldsymbol{p} + \boldsymbol{p} \times \boldsymbol{q}$ (3)

where $p \cdot q$ and $p \times q$ represent the standard inner and cross products, respectively. Meanwhile, $(p_0q_0 - p \cdot q)$ is the product's scalar part, and $(p_0q + q_0p + p \times q)$ is the vector part.

Assume $p_0 = 0$ and $q_0 = 0$, thus P = p and Q = qare pure quaternions. Then, equation (3) becomes

$$PQ = -p \cdot q + p \times q \tag{4}$$

The norm of the quaternion Q is defined as in equation (5). If the norm is 1, the quaternion is called a unit quaternion

$$\|\mathbf{Q}\| = \sqrt{q_0^2 + |\boldsymbol{q}|^2} = \sqrt{q_0^2 + a^2 + b^2 + c^2}$$
(5)

The quaternion $Q^* = q_0 - q$ is called the conjugate of Q. Hence

$$Q^*Q = QQ^* = ||Q||^2$$
(6)

$$\left(\mathbf{PQ}\right)^* = \mathbf{Q}^* \mathbf{P}^* \tag{7}$$

$$\|PQ\| = \|P\| \|Q\|$$
(8)

An arbitrary quaternion Q can be described as¹⁴

$$\mathbf{Q} = \|\mathbf{Q}\|(\cos\theta + n\sin\theta) \tag{9}$$

where **n** is the unit vector, and $\mathbf{n} = \mathbf{q}/||\mathbf{q}||$, $\cos \theta = q_0/||\mathbf{Q}||$ and $0 \le \theta < \pi$.

Quaternions and rotations

Let *S* be an arbitrary inertial reference frame with *i*, *j* and *k* as basic vectors and \mathbf{R}_s be an arbitrary vector. The unit quaternion Q given in quation (10) can represent a counterclockwise rotation which rotates an angle 2θ around the vector *n* in the coordinate system¹⁴

$$\mathbf{Q} = \cos \theta + \mathbf{n} \sin \theta \tag{10}$$

The resulting vector \mathbf{R}'_{s} can be computed through rotation transformation by equation (11)

$$\mathbf{R}'_{s} = \mathbf{Q}\mathbf{R}_{s}\mathbf{Q}^{*} \tag{11}$$

where R'_s is the rotated vector.

Proof. As shown in Figure 2, the vector p rotates around the line l at an angle 20 to q. In order to obtain q, p can be decomposed into p_{\parallel} which is aligned with l and p_{\perp} which is perpendicular to l. p_{\perp} turns into p'_{\perp} after rotation, and p_{\parallel} remains the same, while e is the unit vector of p_{\parallel} . Hence¹⁵

$$\begin{cases} \boldsymbol{p} = \boldsymbol{p}_{\parallel} + \boldsymbol{p}_{\perp} \\ \boldsymbol{q} = \boldsymbol{p}_{\parallel} + \boldsymbol{p}'_{\perp} \end{cases}$$
(12)
$$\boldsymbol{p}'_{\perp} \boldsymbol{p}_{\perp} = -\boldsymbol{p}_{\perp} \cdot \boldsymbol{p}'_{\perp} - \boldsymbol{p}_{\perp} \times \boldsymbol{p}'_{\perp} \\ = -\|\boldsymbol{p}_{\perp}\|\|\boldsymbol{p}'_{\perp}\|\cos 2\theta - \boldsymbol{e}\|\boldsymbol{p}_{\perp}\|\|\boldsymbol{p}'_{\perp}\|\sin 2\theta$$
(13)

where p_{\perp} , p'_{\perp} and *e* comply with right-handed coordinate system principle. It then has

$$\boldsymbol{p}'_{\perp} = (-\|\boldsymbol{p}_{\perp}\|\|\boldsymbol{p}'_{\perp}\|\cos 2\theta - \boldsymbol{e}\|\boldsymbol{p}_{\perp}\|\|\boldsymbol{p}'_{\perp}\|\sin 2\theta)\boldsymbol{p}_{\perp}^{-1}$$

= (\cos 2\theta + \epsilon \sin 2\theta)\epsilon_{\perp} (14)

It is easy to know that

$$\boldsymbol{p}_{\parallel} = (\boldsymbol{p} \cdot \boldsymbol{e})\boldsymbol{e} = \frac{-1}{2}(\boldsymbol{p}\boldsymbol{e} + \boldsymbol{e}\boldsymbol{p})\boldsymbol{e} = \frac{1}{2}(\boldsymbol{p} - \boldsymbol{e}\boldsymbol{p}\boldsymbol{e})$$
(15)

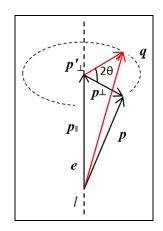


Figure 2. Schematic diagram of rotation.

Table I. Octonion multiplication table.

	I	i	j	k	I	il	jl	kl
I	I	i	i	k	I	il	jl	kl
i	i	- I	, k	—j	il	-1	– kl	jl
j	j	$-\mathbf{k}$	-1	i	jl	kl	-1	_il
k	k	j	—i	- I	kl	—jl	il	-1
1	I	—il	—jl	-kl	- I	i	j	k
il	il	1	-kl	jl	—i	- I	-k	j
jl	jl	kl	1	—il	—j	k	- I	—i
kl	kl	—jl	il	I	-k	—j	i	- I

$$\boldsymbol{p}_{\perp} = \boldsymbol{p} - \boldsymbol{p}_{\parallel} = \frac{1}{2}(\boldsymbol{p} + \boldsymbol{e}\boldsymbol{p}\boldsymbol{e}) \tag{16}$$

Combined with equations (12), (14), (15) and (16), we obtain

$$q = p_{\parallel} + p'_{\perp}$$

= $\frac{1}{2} [(1 + \cos 2\theta)p + \sin 2\theta ep - \sin 2\theta pe + (\cos 2\theta - 1)epe]$
= $(\cos \theta + e \sin \theta)p(\cos \theta - e \sin \theta)$
(17)

which is in accord with equation (11); thus, quaternions can be used to describe the rotations.

Definitions of octonions

Similar to quaternions, the octonions are an eightdimensional (8D) algebra with bases 1, i, j, k, l, il, jland kl, with their multiplications given in Table 1, which gives a detailed description of the multiplying results.¹⁶

It is obvious that the octonions are nonassociative since $(ij)l = -i(jl) = kl \neq i(jl)$. An arbitrary octonion can be described as in equation (18)

$$\mathbf{O} = \mathbf{a}_0 + \mathbf{a}_1 \mathbf{i} + \mathbf{a}_2 \mathbf{j} + \mathbf{a}_3 \mathbf{k} + \mathbf{a}_4 \mathbf{l} + \mathbf{a}_5 \mathbf{i} \mathbf{l} + \mathbf{a}_6 \mathbf{j} \mathbf{l} + \mathbf{a}_7 \mathbf{k} \mathbf{l}$$
(18)

where $a_0 - a_7$ are real numbers.

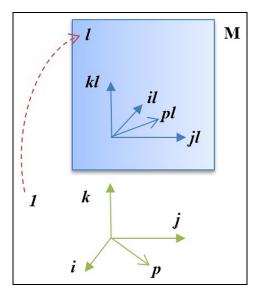


Figure 3. Octonion description by employing a mirror.

Suppose Q_4 is a quaternion and its conjugate Q_4^* . Defining an octonion O_8 , it is easy to find that

$$Q_4 O_8 Q_4^* = (Q_4 O_8) Q_4^* = Q_4 (O_8 Q_4^*)$$
(19)

The first four items of octonions in equation (18) are just a quaternion, which can represent a spatial position rotation in a three-dimensional (3D) space.

The octonion's space can be denoted by the quaternion's space combined with a mirror \mathbf{M} , as shown in Figure 3.

In Figure 3, 1, *i*, *j* and *k* are the four bases of quaternions (*1-space*), and *il*, *jl* and *kl* are the images of *i*, *j* and *k* in the mirror **M**, which comply with the lefthand coordinate system. Hence, *l*, *il*, *jl* and *kl* constitute another four-dimensional (4D) space (*l-space*). As a result, 1, *i*, *j* and *k* and *l*, *il*, *jl* and *kl* constitute an 8D space, which are the bases of octonions. *p* is an arbitrary vector in the *1-space*, and its image in M is *pl*. Similar to pure quaternion, it can be denoted by pure octonion as $O_P = p + pl$. It should be noted that this is only one of the various descriptions for octonions suitable for this article.

Rotations and octonions

Since the octonions were discovered independently by Arthur Cayley in 1845, one of the most familiar applications is to describe the process whereby an electron emits or absorbs a photon,¹⁶ where it can be used to describe the spinor.

Octonions in 3D coordinate system

In Figure 2, the resulting vector $\boldsymbol{q} = \boldsymbol{p}_{\parallel} + \boldsymbol{p}'_{\perp} \cdot \boldsymbol{p}_{\parallel}$ is not affected by the rotation, and it remains unchanged after rotation.

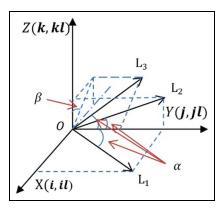


Figure 4. 3D coordinate system.

In Figure 4, assume X, Y and Z axes are rotated around the Z axis by an angle of 2θ , then we obtain

X axis :
$$O'_x = Q_z O_x Q_z^*$$

= $(i \cos 2\theta + j \sin 2\theta) + il(\cos 2\theta - k \sin 2\theta)$ (20)

Y axis :
$$O'_y = Q_z O_y Q_z^*$$

= $(-i\sin 2\theta + j\cos 2\theta) + jl(\cos 2\theta - k\sin 2\theta)$ (21)

Zaxis:
$$O'_z = Q_z O_z Q_z^* = \mathbf{k} + \mathbf{k} \mathbf{l} (\cos 2\theta - \mathbf{k} \sin 2\theta)$$
(22)

where $Q_z = \cos \theta + k \sin \theta$, $Q_z^* = \cos \theta - k \sin \theta$, $O_x = i + il$, $O_y = j + jl$ and $O_z = k + kl$.

It is obvious that $(i\cos 2\theta + j\sin 2\theta)$, $(-i\sin 2\theta + j\cos 2\theta)$ and k represent the spatial position vectors, while $il(\cos 2\theta - k\sin 2\theta)$, $jl(\cos 2\theta - k\sin 2\theta)$ and $kl(\cos 2\theta - k\sin 2\theta)$ describe the rotation processes.

Suppose three lines L_1 , L_2 and L_3 are in the planes XOY and YOZ and the space OXYZ, respectively, with the same inclined angle α with Y axis as shown in Figure 4. The vectors of the three lines can be described as

$$\mathbf{L}_1: \, \boldsymbol{V}_1 = \boldsymbol{i} \sin \alpha + \boldsymbol{j} \cos \alpha \tag{23}$$

$$\mathbf{L}_2: \ \boldsymbol{V}_2 = \boldsymbol{j}\cos\alpha + \boldsymbol{k}\sin\alpha \tag{24}$$

$$L_3: V_3 = -i\sin\alpha\sin\beta + k\sin\alpha\cos\beta + j\cos\alpha$$
(25)

where β is the angle between the Z axis and the projection of L₃ in the plane XOZ.

Define an octonion

$$O = O_p + O_r \tag{26}$$

where $O_p = a_0 + a_1 i + a_2 j + a_3 k$, denoting the position vector, $O_r = a_4 l + a_5 i l + a_6 j l + a_7 k l$, denoting the rotation angle vector. Since the position vector is the same as the quaternions described above during the rotation, here we only focus on the rotation angle vector.

Suppose that Y axis is rotated 2θ around L_1 , L_2 and L_3 , respectively, then we obtain

$$L_{1}: O_{r1} = Q_{l1}O_{ry}Q_{l1}^{*}$$

$$= -l(\sin 2\theta \cos \alpha + j\cos 2\theta - k\sin 2\theta \sin \alpha)$$

$$(27)$$

$$L_{2}: O_{r2} = Q_{l2}O_{ry}Q_{l2}^{*}$$

$$= -l(\sin 2\theta \cos \alpha + j\cos 2\theta + i\sin 2\theta \sin \alpha)$$
(28)

$$O_{r3} = Q_{l3}O_{ry}Q_{l3}$$

= $-I[\sin 2\theta \cos \alpha + j \cos 2\theta + (i \sin 2\theta \sin \alpha \cos \beta + k \sin 2\theta \sin \alpha \sin \beta)]$
(29)

where $Q_{l1} = \cos \theta + V_1 \sin \theta$, $Q_{l1}^* = \cos \theta - V_1 \sin \theta$, $Q_{l2} = \cos \theta + V_2 \sin \theta$, $Q_{l2}^* = \cos \theta - V_2 \sin \theta$, $Q_{l3} = \cos \theta + V_3 \sin \theta$, $Q_{l3}^* = \cos \theta - V_3 \sin \theta$ and $O_{ry} = jl$.

The results indicate that $(\sin 2\theta \cos \alpha)$ is a constant; ($j \cos 2\theta$) is the segment related to the rotated vector (Y axis); $(-k \sin 2\theta \sin \alpha)$, ($i \sin 2\theta \sin \alpha$) and $[(i \cos \beta + k \sin \beta) \sin 2\theta \sin \alpha]$ are the segments describing the perpendicular vector that complies with the left-hand coordinate system with the rotation axis and Y axis. Combined with equations (20), (21) and (22), the rotation process of a vector around an axis can be denoted by two rotation segments of an octonion, that is, the rotating component around a perpendicular axis and the spinor, the coefficients are invariant in different coordinate systems and are determined by the rotation quaternion.

Arbitrary nonorthogonal rotation

An arbitrary nonorthogonal rotation can be shown in Figure 5, \overrightarrow{OA} makes an inclined angle α ($0 < \alpha < \pi/2$) with OP and rotates 2θ ($0 \le 2\theta < \pi$) around OP to \overrightarrow{OB} . Define the original vector OA as OY1 axis; OY₀, OY₁, OZ₀ and OZ₁ are in the same plane; OZ₂ is perpendicular to \overrightarrow{OA} and \overrightarrow{OB} ; and $OX_0 \perp OY_0 \perp$ OZ_0 , $OX_0 \perp OY_1 \perp OZ_1$, $OX_2 \perp OY_1 \perp OZ_2$ and $\overrightarrow{AM} \perp \overrightarrow{OA} \perp OX_0$, \overrightarrow{AM} could be used to denote the spinning angle of $\overrightarrow{O} A$.

In the coordinate system $OX_0Y_1Z_1$, the rotation is same as the rotation around L_2 in Figure 4. Take $\overline{O} A$ and $\overline{O} B$ as unit vectors, the result can be described as

$$O_{OB1} = Q_{OP}O_{OA}Q_{OP}^{*}$$

= $-l\sin 2\theta\cos\alpha + jl\cos 2\theta + il\sin 2\theta\sin\alpha$ (30)

where $Q_{OP} = \cos \theta + (j \cos \alpha + k \sin \alpha) \sin \theta$ and $Q_{OP}^* = \cos \theta - (j \cos \alpha + k \sin \alpha) \sin \theta$ and $O_{OA} = jl$.

Meanwhile, in the coordinate system $OX_2Y_1Z_2$, the rotation can be divided into rotating around OZ_2 which represents the position rotation process and spinning around itself. As $\overline{O} A$ rotates 2θ around $\overline{O} P$, based on two rotation quaternions, $(\cos \gamma + \mathbf{k} \sin \gamma)$ and $(\cos \delta + \mathbf{j} \sin \delta)$, the combined quaternion vector is defined as

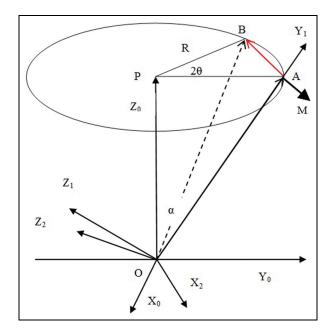


Figure 5. Arbitrary nonorthogonal rotation.

 $\mathbf{Q}_0 = \cos \theta + \mathbf{k} \sin \gamma + \mathbf{j} \sin \delta \tag{31}$

where 2γ is the rotating angle (around OZ₂) and 2δ is the spinning angle (around OY₁) in the coordinate system OX₂Y₁Z₂. It can be shown $\sin^2\gamma + \sin^2\delta = \sin^2\theta$. Then, we obtain

$$O_{OB2} = Q_0 O_{OA} Q_0^*$$

= $-I2 \sin \delta \cos \theta + j l \cos 2\theta + i l 2 \sin \gamma \cos \theta$
(32)

From $O_{OB1} = O_{OB2}$, we obtain

$$\begin{cases} 2\delta = 2\sin^{-1}(\sin\theta\cos\alpha)\\ 2\gamma = 2\sin^{-1}(\sin\theta\sin\alpha) \end{cases} \left(0 \le \alpha \le \frac{\pi}{2}, 0 \le \theta, \delta, \gamma < \pi \right) \end{cases}$$
(33)

According to Figure 5, it is facile to find that 2γ is $\angle AOB$, and 2δ is the rotation angle of AM around $\overline{O}A$; since OZ_2 is not parallel to AM, in the rotating process of $\overline{O}A$ around OZ_2 , AM would produce minor rotation around $\overline{O}A$, which means 2δ is not the real spinning angle of $\overline{O}A$. However, as the inclined angle of AM and OZ_2 is minute, this article chooses 2δ as the spinning angle of $\overline{O}A$, and the minor error is hence neglected.

Calibration error analysis

In section "Rotations and octonions," the rotation has been realized by means of decomposing the axes, compared to the decomposition of the rotated vector described in section "Quaternions and octonions." In Figure 6, M_N is the surface axis (surface normal) of the reflecting mirror, M_H the horizontal axis and M_V the vertical axis. P_A and P_E are the azimuth axis and the elevation axis of the platform, respectively. T_A and T_E are the medium axis and the outer axis of the turntable,

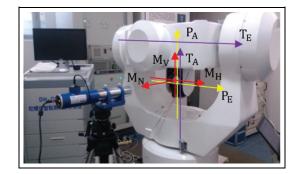


Figure 6. Experimental setup and data acquisition system.

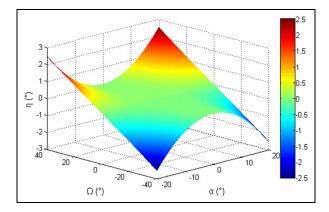


Figure 7. Calibration error's effect.

respectively. At zero positions, P_A and T_A are vertical, and P_E and T_E are horizontal.

As described in section "Introduction," the azimuth and elevation angles are defined for the platform (θ_{PA} and θ_{PE}), the turntable (θ_{TA} and θ_{TE}) and the autocollimator readouts of axes (δ_A and δ_E). When $\theta_{TE} = -\theta_{PE} = \alpha \neq 0$, if the turntable rotates Ω around T_A, the mirror will rotate at the same angle around M_V, if the platform continues rotating $-\Omega$ around P_A, according to equation (33), the mirror will only rotate $-[2\sin^{-1}(\sin(\Omega/2)\cos\alpha)]$ around M_V with a spinor of $-[2\sin^{-1}(\sin(\Omega/2)\sin\alpha)]$ around M_H. The spinning of the mirror will result in the coupling error in the azimuth and elevation, but the angle is minute in this calibration process. The rotation angle will introduce an angular error η in the azimuth

$$\eta = \Omega - \left[2\sin^{-1}\left(\sin\frac{\Omega}{2}\cos\alpha\right)\right]$$
(34)

With Ω changing from -40° to 40° and α from -20° to 20° , the calibration error in simulation is shown in Figure 7. Since the calibration error significantly increases with Ω and α , it is necessary to separate the error.

Experimental results

The pointing errors have been acquired with a highprecision turntable and an autocollimator at the

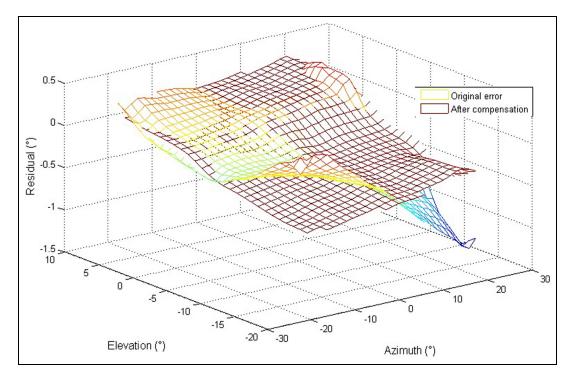


Figure 8. Azimuth error before and after compensation.

Table 2. Compensation result comparisons.

	Azimuth (original error)	Azimuth (compensated)
Mean value	-0.1703°	−0.1674 °
Variance	0.1379(°) ²	0.0492(°) ²
Maximum value	0.6935°	0.2496 [`]
Minimum value	-1.5610°	$-$ 0.5880 $^{\circ}$

azimuth of -20° , -15° , -10° , -5° , 0° , 5° , 10° , 15° and 20° and at the elevation of -20° , -15° , -10° , -5° , 0° , 5° and 10° . The compensation results calculated by equation (34) (the elevation angle is α , and the azimuth angle is the rotated angle Ω) are shown in Figure 8 and Table 2, which have shown that the calibration error has been compensated effectively. However, in order to improve its pointing accuracy further, it is necessary to analyze errors that result from the platform's misalignment error, nonperpendicularity, initialization error and so on, and then relevant compensation models can be adopted. The detailed information has been presented and reported in Hong et al.⁹ and Tang et al.¹⁷

Conclusion

With respect to the ISPs, calibration process with a high-precision turntable and an autocollimator, this article has developed a novel approach to the rotation decomposition to determine the calibration errors. It has shown that octonions can represent the rotation process completely, including the rotation around a space vector and the spinning around itself. It has realized a new representation of the rotation by means of decomposing the rotation axis into two perpendicular axes. The error in the calibration process is then derived. According to the test results, the accuracy after error correction is significantly improved with the variance in azimuth decreased from $0.1379(^{\circ})^2$ to $0.0492(^{\circ})^2$, which offers a more accurate set of data for further compensation based on the error model of the platform itself.

Declaration of conflicting interests

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References

- Evans PR. Rotations and rotation matrices. Acta Crystallogr 2001; 57: 1355–1359.
- Fillmore JP. A note on rotation matrices. *IEEE Comput* Graph 1984; 4: 30–33.
- Insel AJ. Rotation matrices in the plane without trigonometry. *Coll Math J* 1993; 24: 71–73.

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- 4. Ozgoren MK. Kinematic analysis of spatial mechanical systems using exponential rotation matrices. *J Mech Des: T ASME* 2007; 129: 1144–1152.
- 5. Neef J. Rotation matrices. Mach Des 1988; 60: 100–102.
- 6. Hilkert JM. Inertially stabilized platform technology. *IEEE Contr Syst Mag* 2008; 28: 26–46.
- 7. Michael KM. Inertially stabilized platforms for optical imaging systems. *IEEE Contr Syst Mag* 2008; 28: 47–64.
- 8. Ekstrand B. Equations of motion for a two-axes gimbal system. *IEEE T Aero Elec Sys* 2001; 37: 1083–1091.
- Hong HJ, Zhou XY, Zhang ZY, et al. Modelling and calibration of pointing errors using a semi-parametric regression method with applications in inertially stabilized platforms. *Proc IMechE, Part B: J Engineering Manufacture*. Epub ahead of print 16 August 2013. DOI: 10.1177/0954405413475559.
- Zhang ZY, Zhou XY and Fan DP. Analysis, modelling and correction of pointing errors for electro-optical detection systems. *Acta Aeronaut Astronaut Sin* 2011; 32: 2042–2054.

- 11. Rue AK. Calibration of precision gimbaled pointing systems. *IEEE T Aero Elec Sys* 1970; 6: 697–706.
- Goldman R. Understanding quaternions. *Graph Model* 2011; 73: 21–49.
- Meister L and Schaeben H. A concise quaternion geometry of rotations. *Math Method Appl Sci* 2005; 28: 101–126.
- 14. Meister L. Mathematical modelling in geodesy based on quaternion algebra. *Phys Chem Earth Pt A* 2000; 25: 661–665.
- 15. Liu JF. Three dimensional rotation represented by quaternion. *College Phys* 2004; 23: 39–43.
- 16. Baez JC. The octonions. Bull Am Math Soc 2001; 39: 145–205.
- Tang QJ, Wang XJ and Yang Q. Static pointing error analysis of electro-optical detection systems. *Proc IMechE, Part B: J Engineering Manufacture*. Epub ahead of print 15 October 2014. DOI: 10.1177/ 0954405414551107.